VOLUME II JANUARY 1963

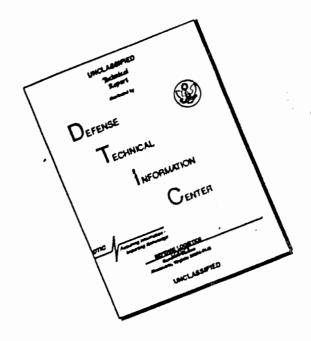
53-5B PROPULSION SYSTEM GHTWORTHINESS TEST REPORT



CONTRACT NUMBER DA44-177-TC-715

GENERAL & ELECTRIC

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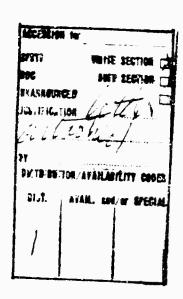
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LIFT FAN FLIGHT RESEARCH AIRCRAFT PROGRAM

Contract DA 44-177-TC-715

X353-5B PROPULSION SYSTEM FLIGHTWORTHINESS TEST REPORT

VOLUME II

January, 1963



GENERAL ELECTRIC COMPANY

FLIGHT PROPULSION LABORATORY DEPARTMENT

CINCINNATI 15, OHIO



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INTRODUCTION

This volume of the FWT report presents photographs with a brief identification of hardware condition and discrepancies found after completion of tests described in Specifications number 114 and 115.

Certificates of inspection are included. Clearance checks were identical with origin: 1 assembly values.

Discussion and recommendations are included in Volume I.

HARDWARE REVIEW AFTER FWT

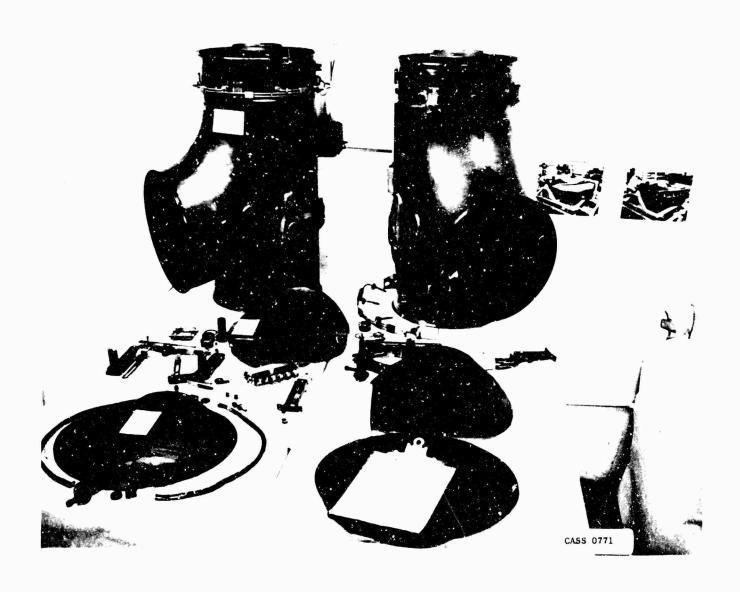


Figure II-1. Hardware Review
Diverter Valve Parts (CASS 0771)



Figure II-2. Hardware Review
Lift Fan Parts (CASS 0767)

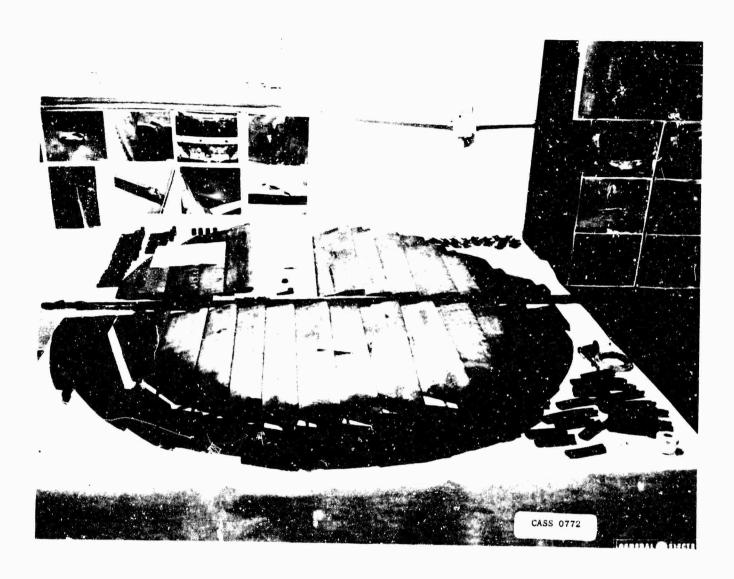


Figure II-3. Hardware Review
Iift Fan Parts (CASS 0772)

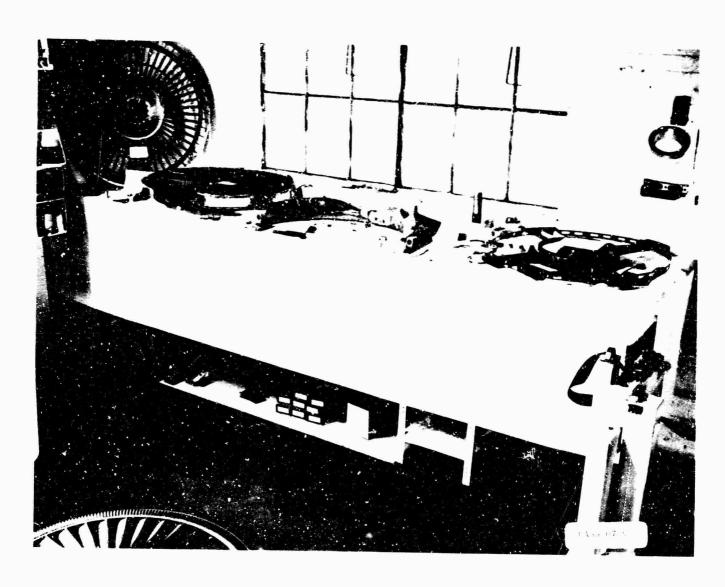


Figure II-4. Hardware Review Pitch Fan Parts (CASS 0766)



Figure II-5. Hardware Review

FWT Hardware Display After Cleaning and Inspection (CASS 0773)



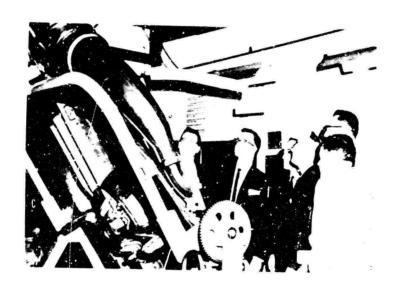
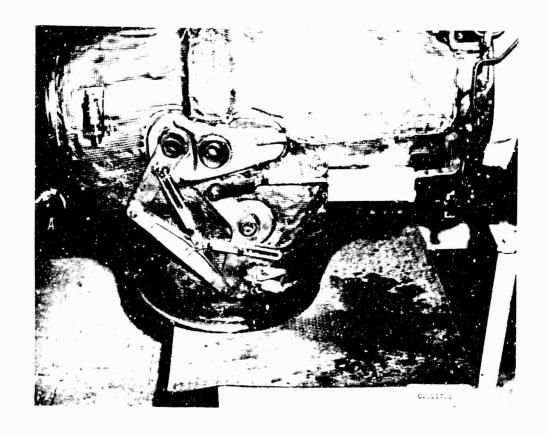


Figure II-6. Hardware Review
A.B.&C. ARMY/FAA Personnel Reviewing Hardware
Condition and Inspection Results (CASS 0780, 0777, 0779)

DISCREPANCIES
DIVERTER VALVE



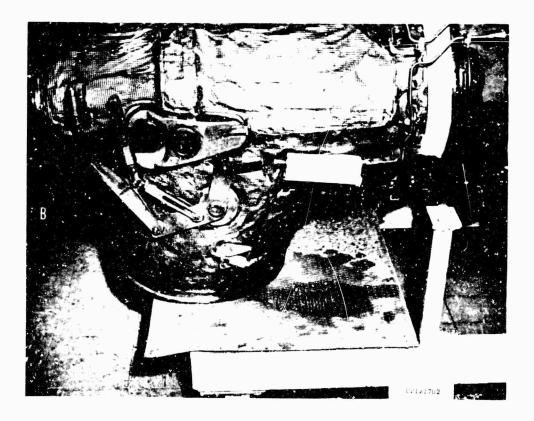


Figure II-7. Diverter Valve Assembly
Diverter Valve After Completing FWT
A. Buckled Push Rod (Lift Mode) (C2121701)
B. Buckled Push Rod (Cruise Mode) (C2121702)
Note: Actuation Not Official FWT Part

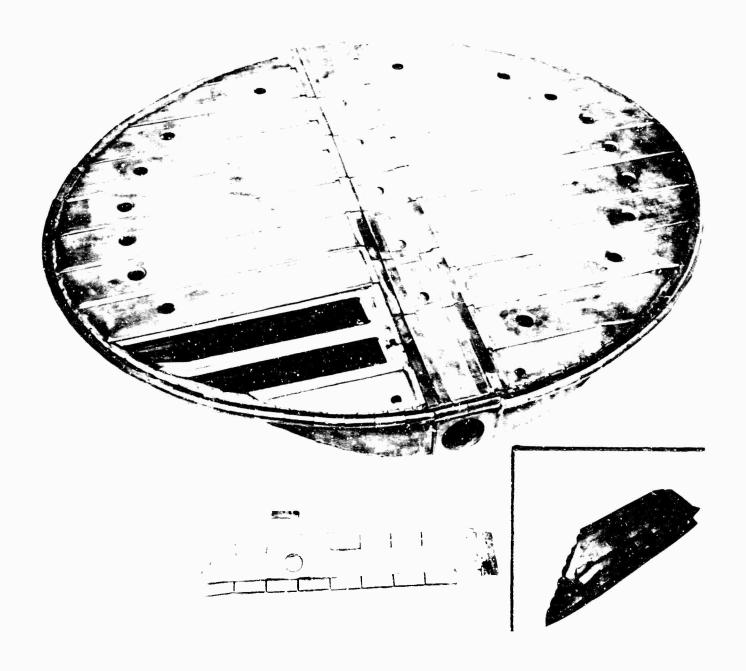


Figure II-8. Door

Heat Shield Failure (Buckled Into Air Stream Because of Insufficient Assembly Clearance to Permit Thermal Growth)

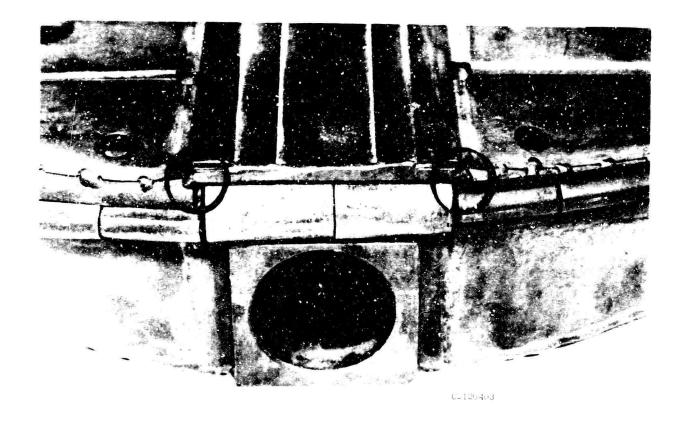


Figure II-9. Doors
Thermal Weld Cracks (C 2120403)

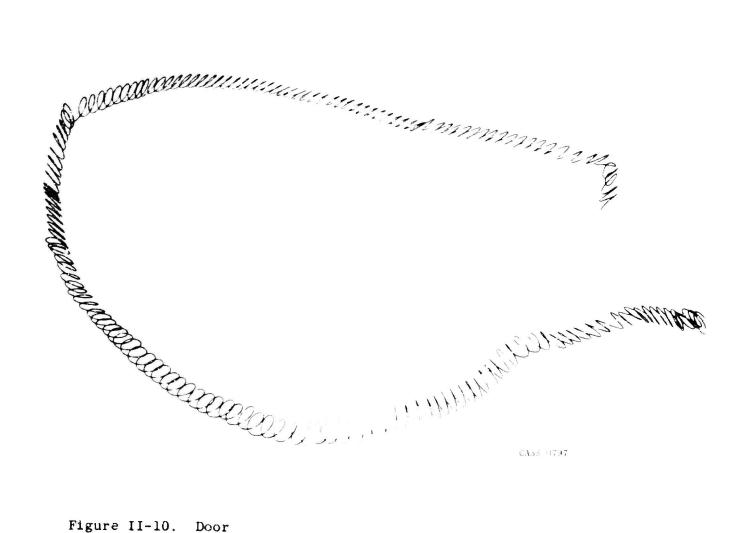


Figure II-10. Door
Seal Spring Deformed By Improper Disassembly (CASS 0797)

DISCREPANCIES LIFT FAN

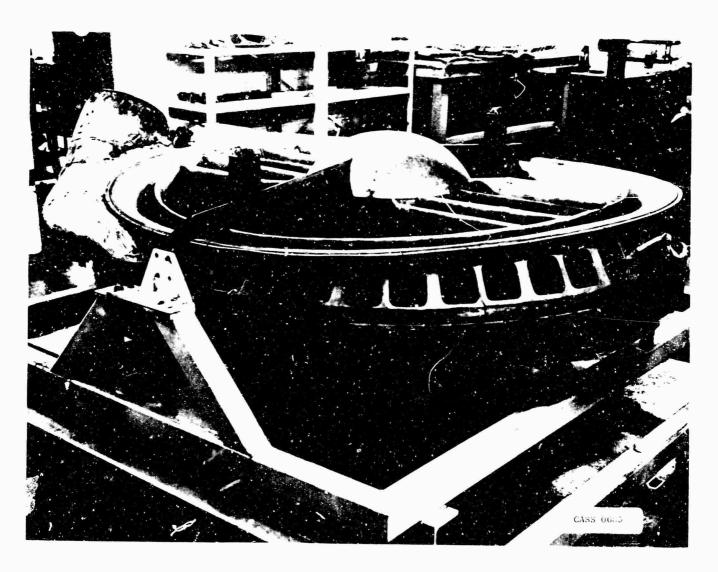
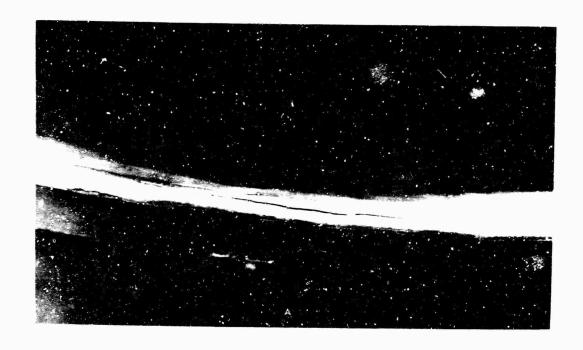
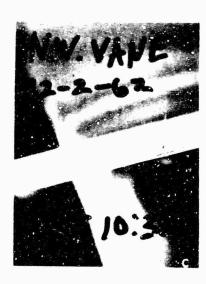


Figure II-11. Lift Fan Assembly
Lift Fan After Completing FWT (CASS 0685)







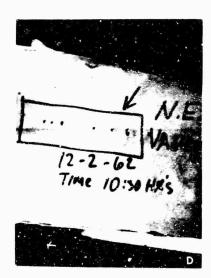
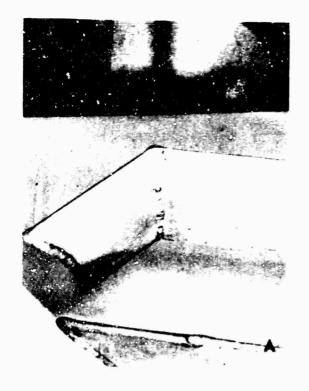
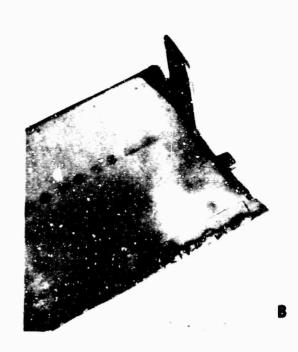


Figure II-12. Inlet Vane

- A. Typical Leading Edge Crack
- B. Typical Trailing Edge Crack
- C. Stop Drill Repair Leading Edge
- D. Typical Spot Weld Cracking; Stop Drill Repair (Note Propagation) (C 2111534 plus 3 Polaroids)





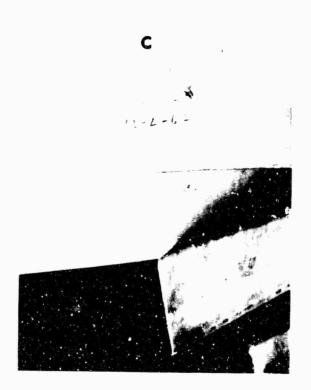
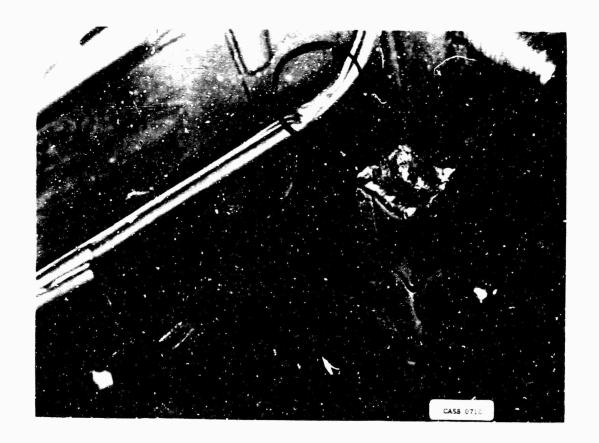


Figure II-13. Inlet Vane Repairs

- A. Leading Edge Welds
- B. Trailing Edge Welds
 C. Spot Weld Repair (3 Polaroids)



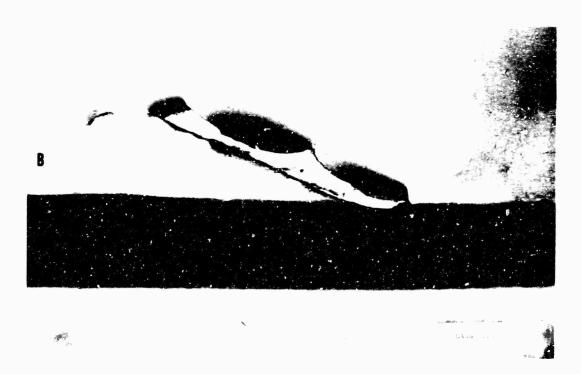


Figure II-14. Front Frame

- A. Dent In Corner Scroll Gas Seal (CASS 0718)
- B. Tear In Internal Insulation Blanket (CASS 0720) Probable Cause: Migrating Carrier Bolt Head

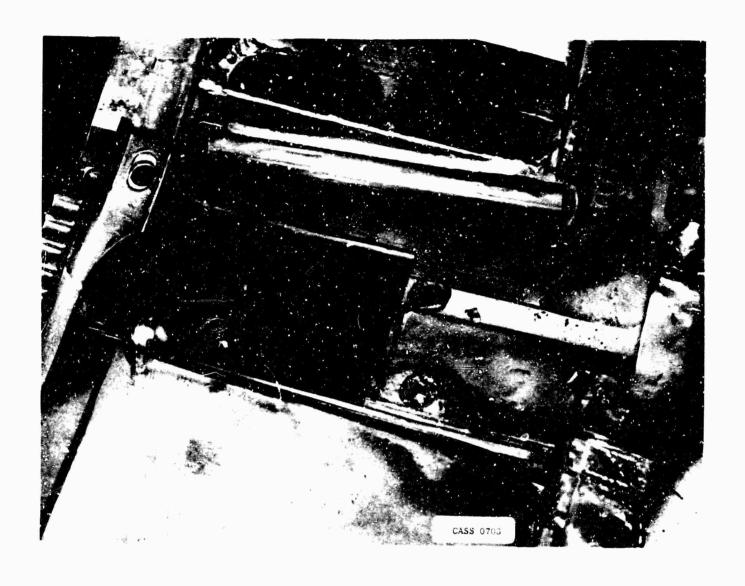


Figure II-15. Front Frame
Failed Carrier Bolt Head From Aft* Side Of Rotor Found
At Disassembly Lodged Under Edge Of Scroll Gas Seal
* Determined At Metallographic Inspection (CASS 0708)



Figure II-16. Scroll
Finger Seal Slipped Off Scroll Bead
(No Gas Leakage) (CASS 0716)



Figure II-17. Scroll

Dent In Nozzle

Probable Cause: Bolt Head Figure II-22 (CASS 0719)

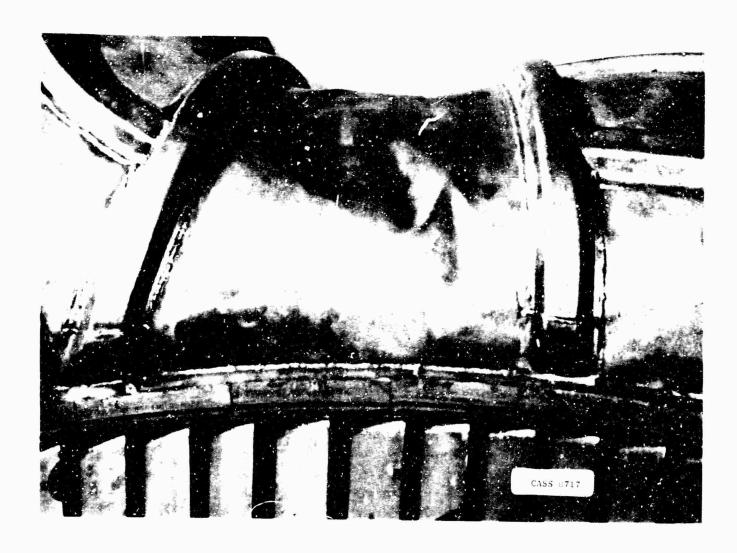


Figure II-18. Scroll
Thermal Buckle In Scroll Divider Region (CASS 0717)

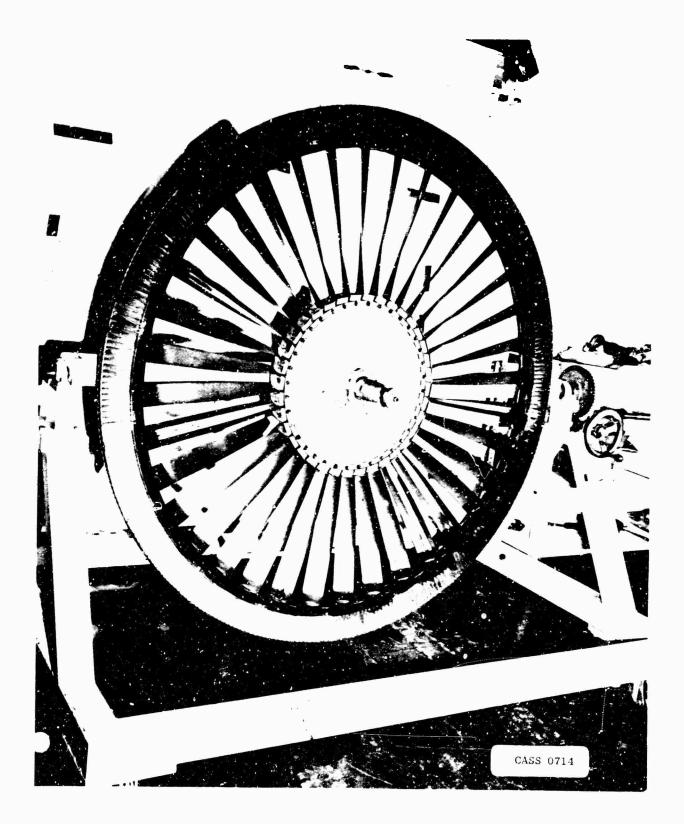


Figure II-19. Rotor
Rotor Sub-Assembly Prior To Disassembly
After Completing FWT (CASS 0714)

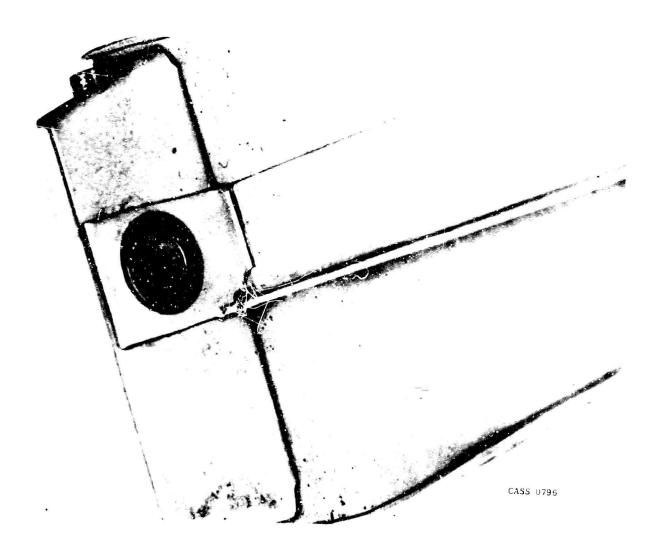


Figure II-20. Rotor
Fatigue Crack In Blade Platform
Cause: Crack Initiated At Sharp Corner Of Groove
Machined In Platform Stifferer To Provide
Clearance For The Rotor Retainer Ring (CASS 0796)

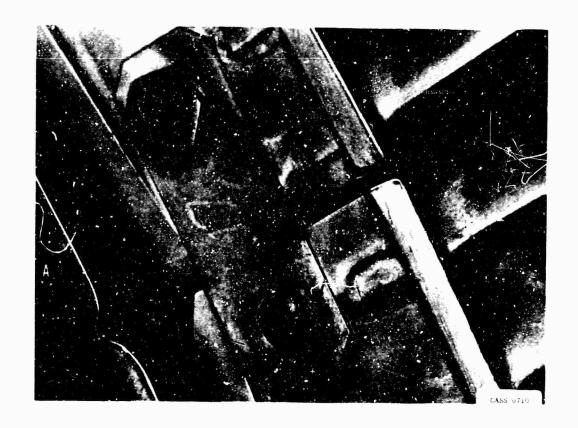




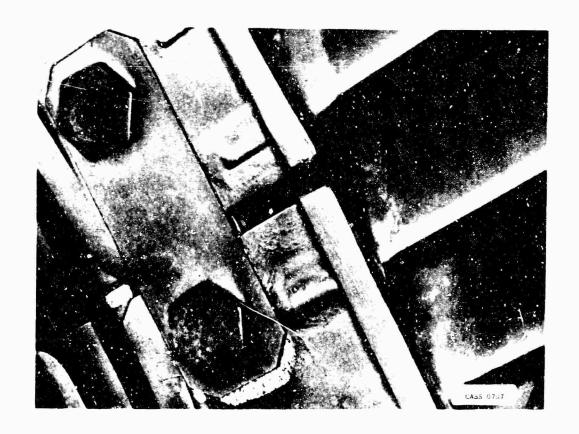
Figure II-21. Rotor

- A. Carrier Bolt Head Missing: Forward Side Carrier 25, Adjacent to Carrier 32. (CASS 0710)
- B. Two Carrier Bolt Heads And Tab Missing: Aft Side Between Carriers 13 and 23. (CASS 0706)

Probable Cause: Defective Bolts



Figure II-22. Rotor
(X10 Mag.)
Carrier Bolt Head Found In Front Frame
(See Figure II-15)
Note: From Aft Side Of Rotor Per Metallographic
Inspection



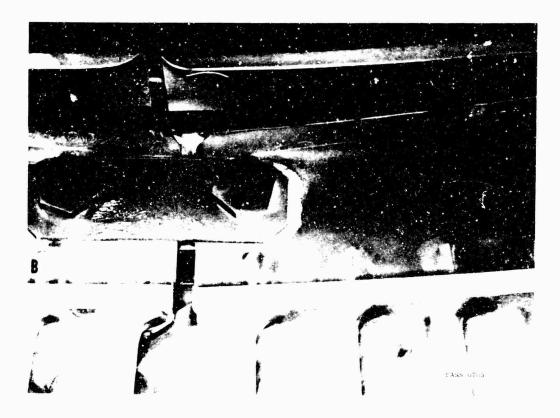


Figure II-23. Rotor

- A. Galling On Carrier 05 (Adjacent To Carrier 07) On Forward Side Of Rotor (CASS 0707)
- B. Galling On Carrier 05 (Adjacent To Carrier 07) On Aft Side Of Rotor (CASS 0705)

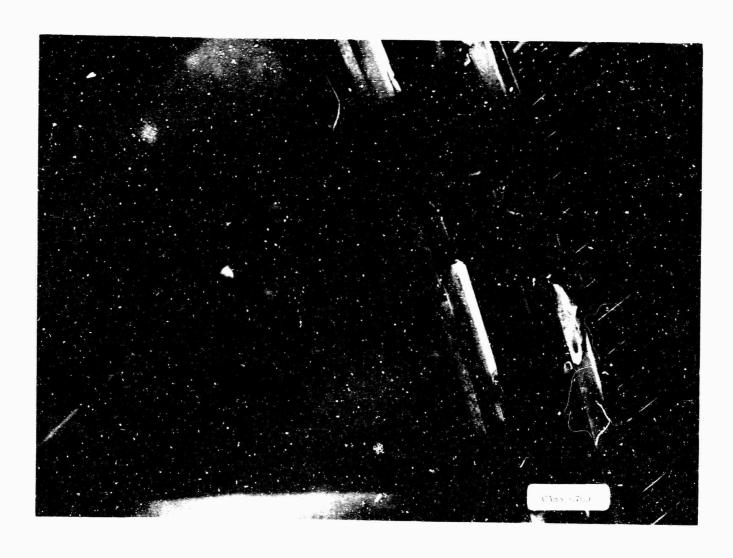


Figure II-24. Rotor
Fretting On Air Seal Carrier 05 (Adjacent To
Carrier 07) (CASS 0709)
Note: Missing Section Of Aft Torque (Buckle Area)
And Opposite Break In Forward Band



CASS 0793

Figure II-25. Rotor
Torque Band Segment
Band Cut During Test As Design Change

w w

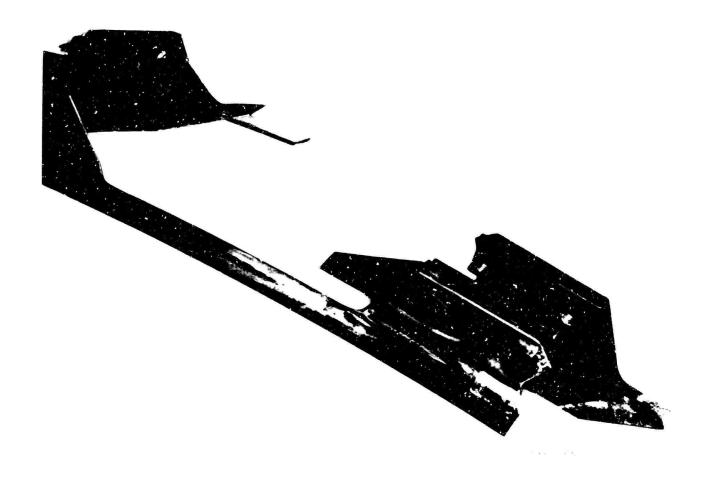


Figure II-26. Rotor
Forward Torque Band Break Opposite Buckle In Aft Band
(Noted After Completion Of 10th Endurance Cycle)
Note: Cover and Band Fretting (CASS 0792)

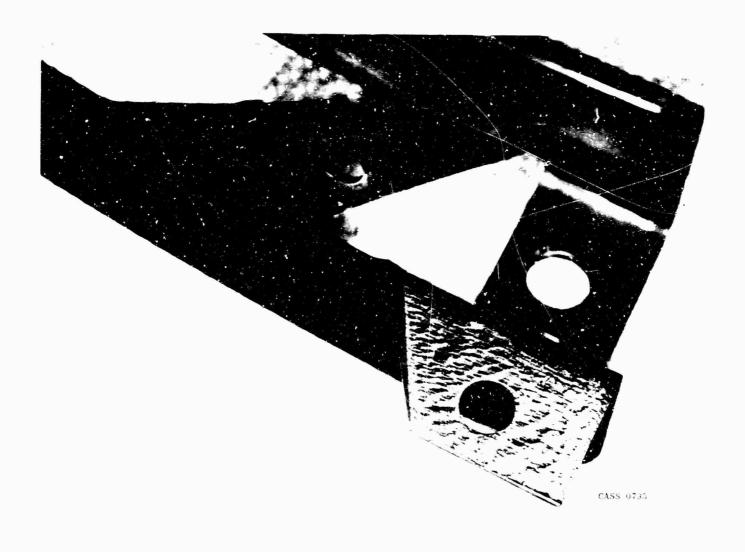


Figure II-27. Rotor
Seal Attachment Ear Fretting (Aft Side Of Rotor)
At Joint Where Forward Band Broke (CASS 0795)

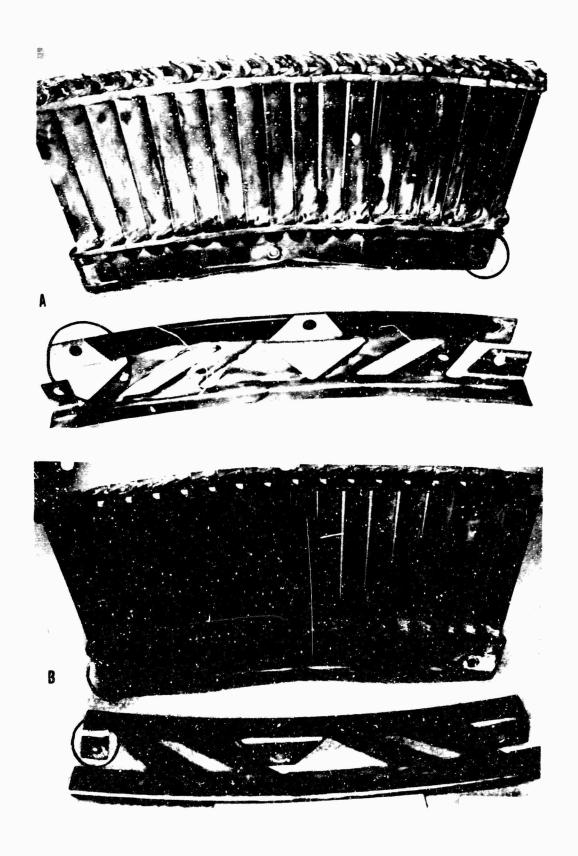


Figure II-28. Rotor
A. Carrier S/N 05 Forward Side (CASS 0722)
B. Carrier S/N 05 Aft Side (CASS 0723)

Note: Seal Galling, Fretting, Airfoil Gouge Physical Damage To All Carriers Caused By Migration Of Carrier Tab (Figure II-46)

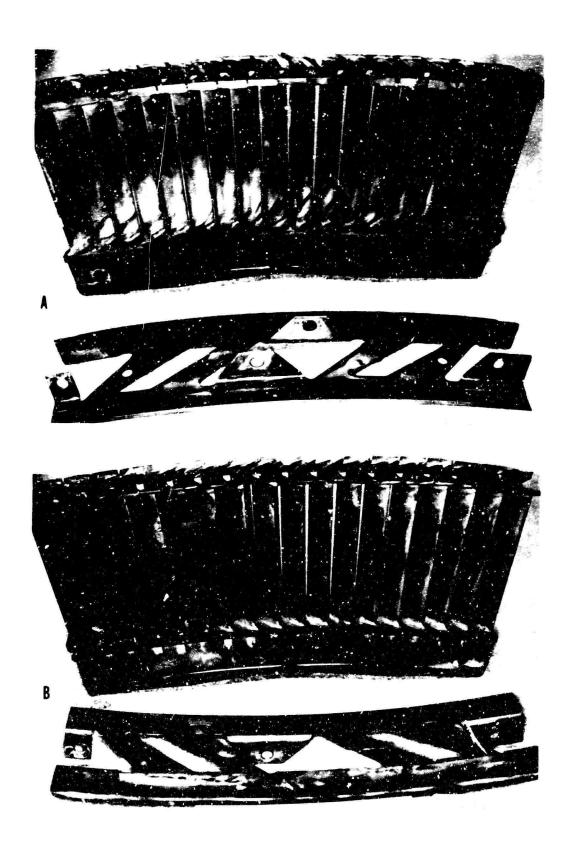


Figure II-29. Rotor
A. Carrier S/N 007 Forward Side (CASS 0725)
B. Carrier S/N 007 Aft Side (CASS 0724)
Note: Shroud Tear, Airfoil Punctures

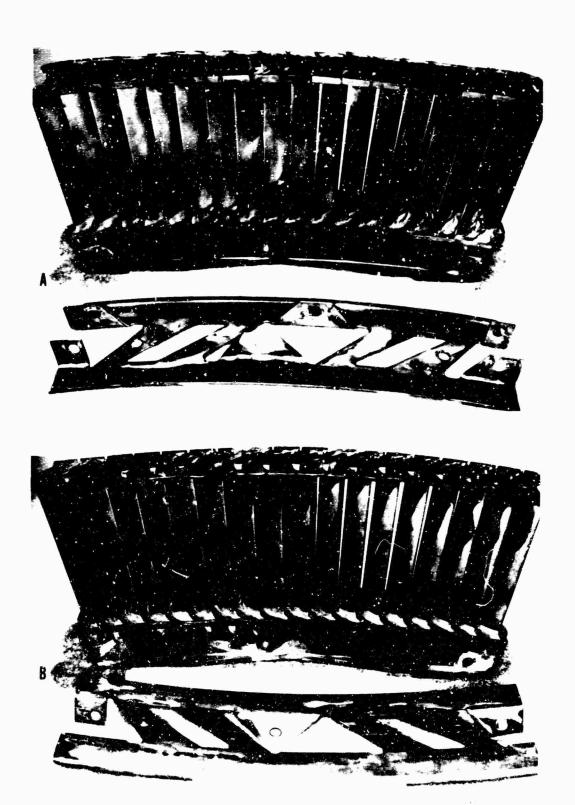


Figure II-30. Rotor
A. Carrier S/N 08 Forward Side (CASS 0750)
B. Carrier S/N 08 Aft Side (CASS 0749)
Note: Fretting

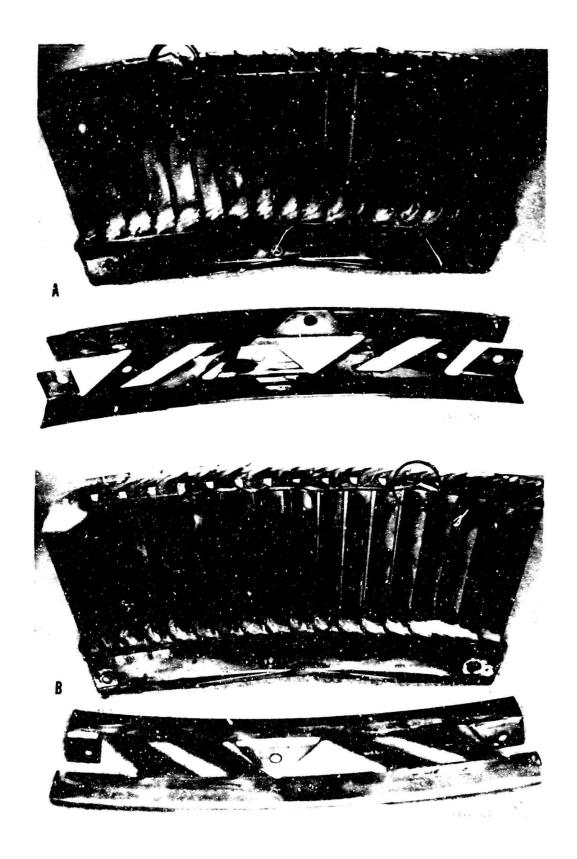
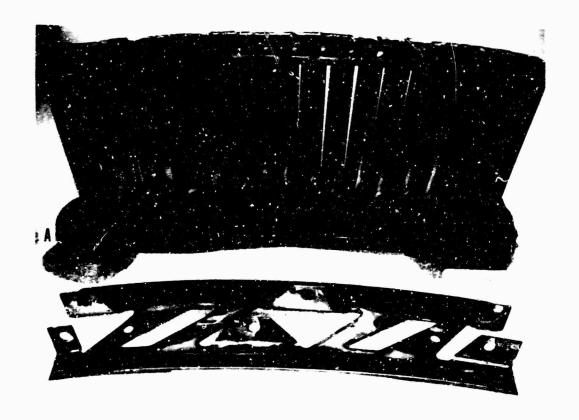


Figure II-31. Rotor
A. Carrier S/N 10 Forward Side (CASS 0726)
B. Carrier S/N 10 Aft Side (CASS 0727)
Note: Shroud Separation, Bent Airfoils



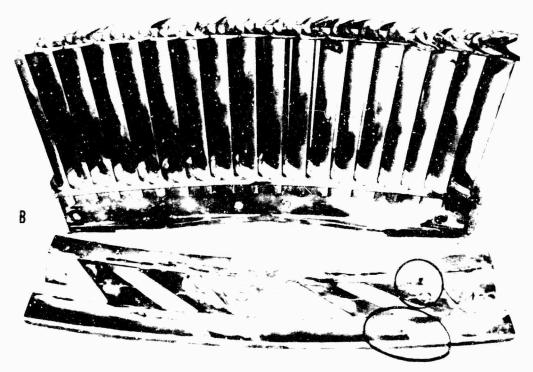


Figure IJ-32. Rotor

A. Carrier S/N L-13 Forward Side (CASS 0746)
B. Carrier S/N L-13 Aft Side (CASS 0763)
Note: Dents, Seal Fretting



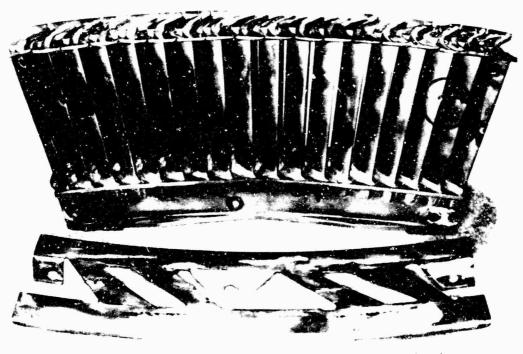


Figure II-33. Rotor

A. Carrier S/N 16 Forward Side (CASS 0747)

B. Carrier S/N 16 Aft Side (CASS 0748)

Note: Airfoil Tears, Dents

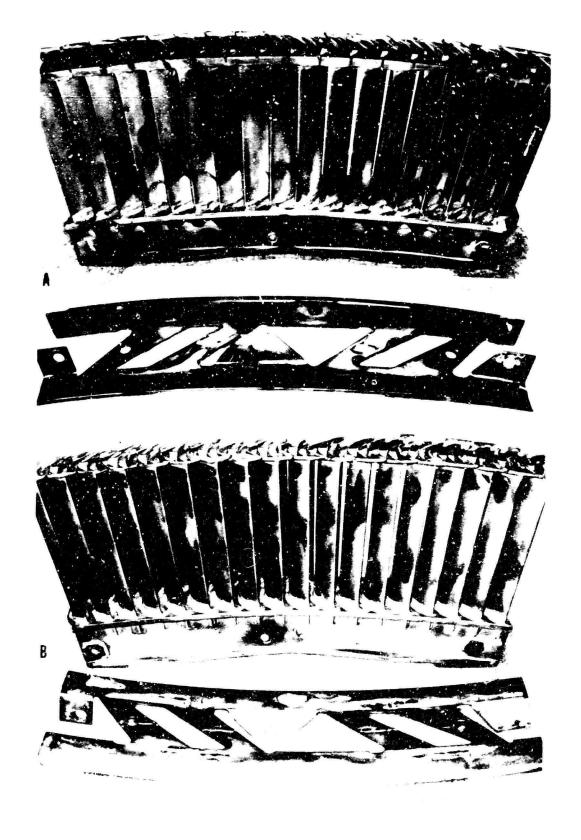
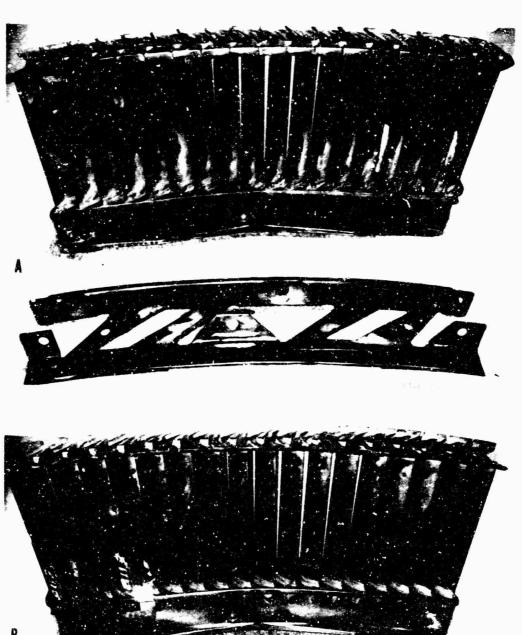


Figure II-34. Rotor

A. Carrier S/N 17 Forward Side (CASS 0729)
B. Carrier S/N 17 Aft Side (CASS 0728)

Note: Shroud Damage



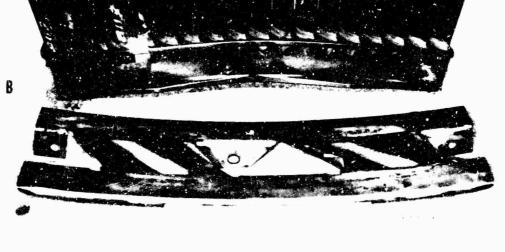


Figure II-35. Rotor

A. Carrier S/N 20 Forward Side (CASS 0733)
B. Carrier S/N 20 Aft Side (CASS 0734)
Note: Airfoil Tear, Puncture, Fretting

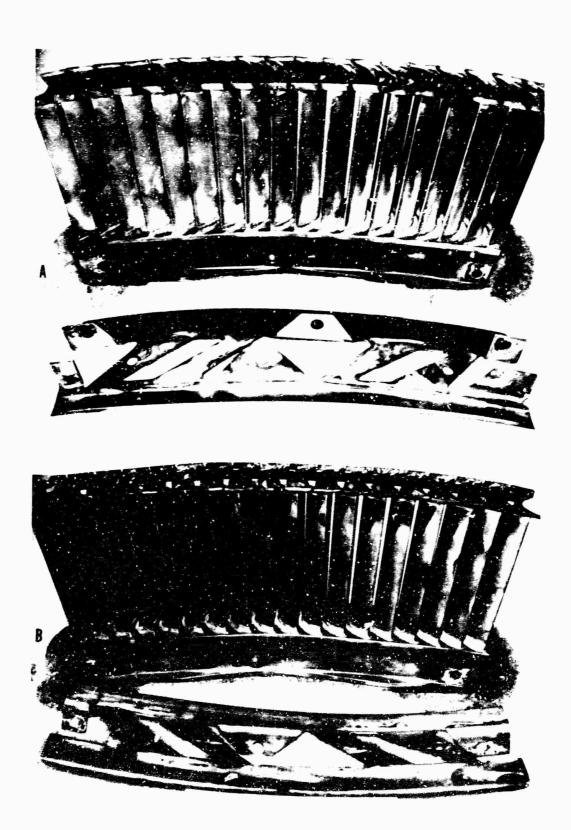


Figure II-36. Rotor

A. Carrier S/N 21 Forward Side (CASS 0741)

B. Carrier S/N 21 Aft Side (CASS 0745)

Note: Airfoil Puncture

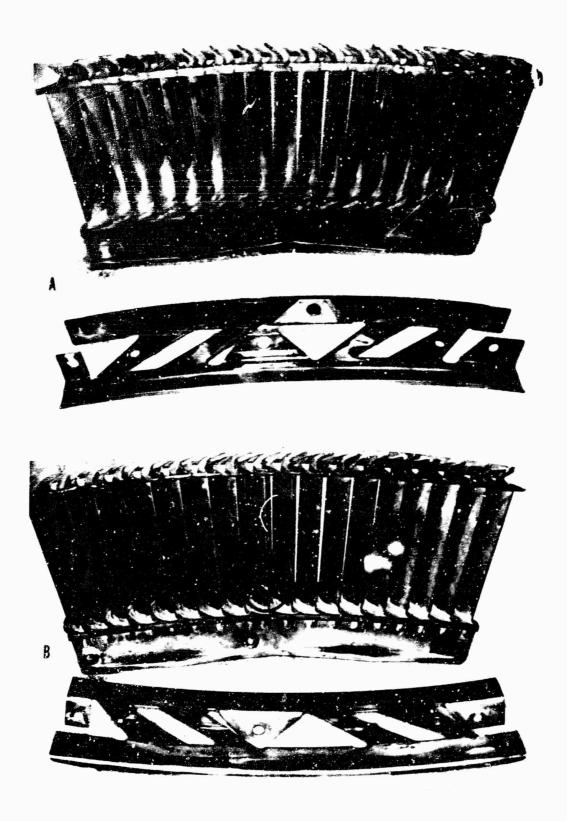


Figure II-37. Rotor

Carrier S/N 22 Forward Eide (CASS 0735)
B. Carrier S/N 22 Aft Side (CASS 0736)

Note: Airfoil Puncture



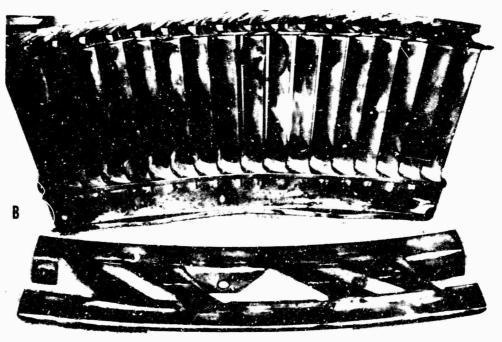


Figure II-38. Rotor

A. Carrier S/N 23 Forward Side (CASS 0755)
B. Carrier S/N 23 Aft Side (CASS 0756)
Note: Airfoil Tears

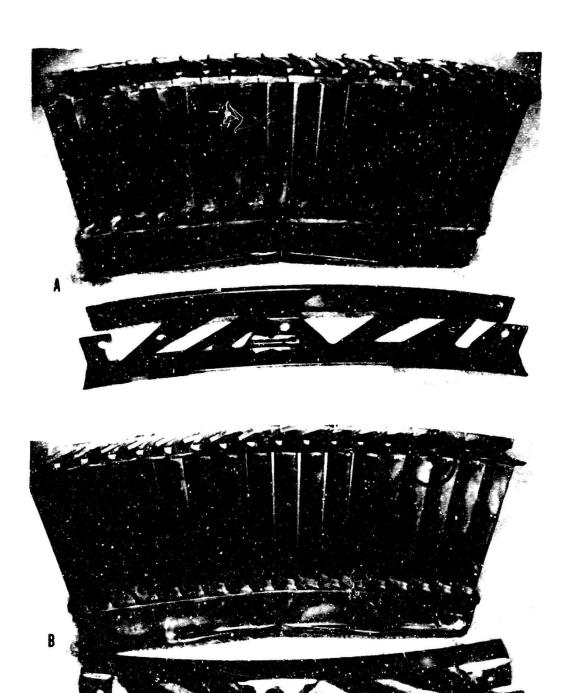


Figure II-39. Rotor

A. Carrier S/N 25 Forward Side (CASS 0737)
B. Carrier S/N 25 Aft Side (CASS 0738)

Note: Airfoil Tear

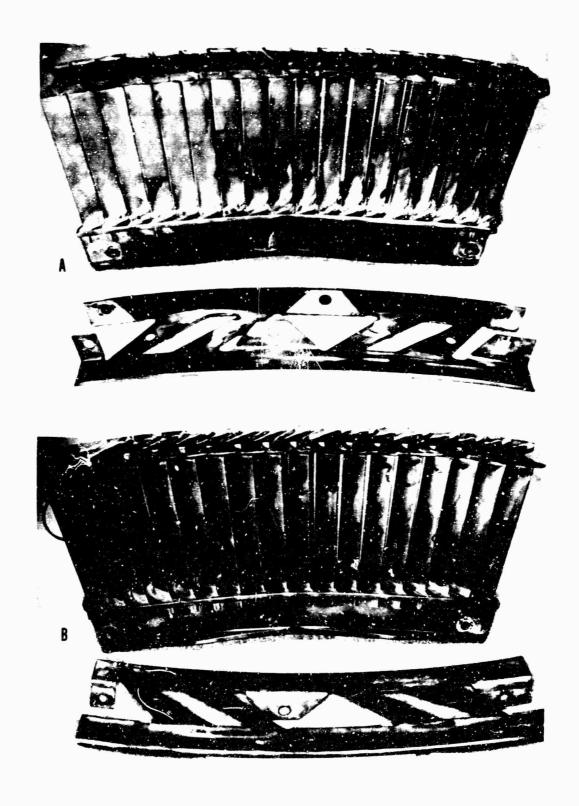


Figure II-40. Rotor

A. Carrier S/N 27 Forward Side (CASS 0739)
B. Carrier S/N 27 Aft Side (CASS 0740)
Note: Airfoil Tear

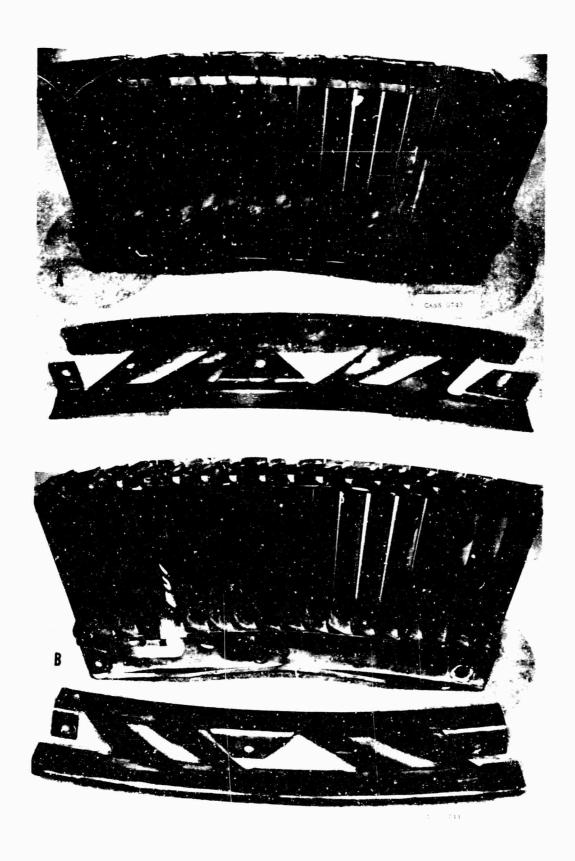


Figure II-41. Roter

A. Carrier S/N 30 Forward Side (CASS 0743)
B. Carrier S/N 30 Aft Side (CASS 0744)
Note: Shroud Tears



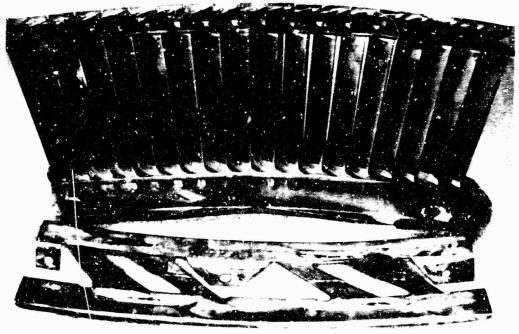


Figure II-42 Rotor

A. Carrier S/N 31 Forward Side (CASS 0732)
B. Carrier S/N 31 Aft Side (CASS 0754)
Note: Shroud Tear

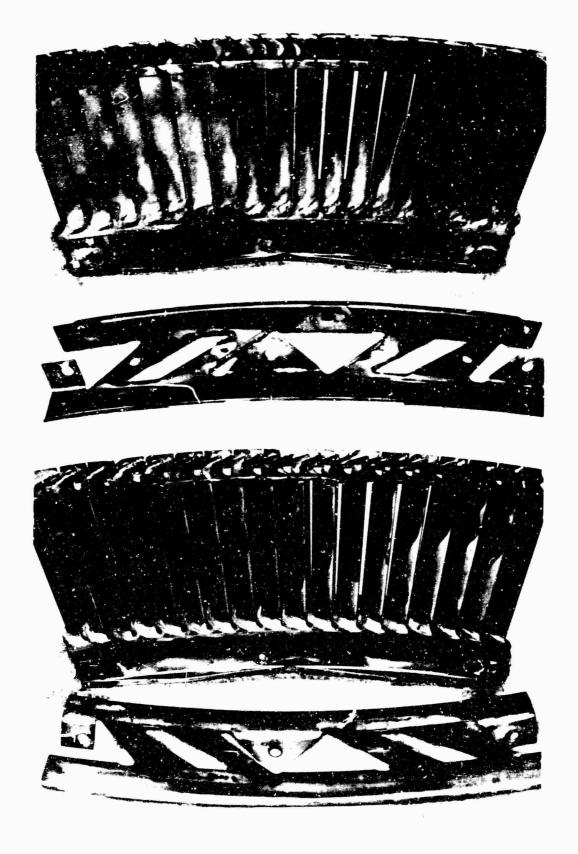


Figure II-43. Rotor
A. Carrier S/N 32 Forward Side (CASS 0730)
B. Carrier S/N 32 Aft Side (CASS 0731)

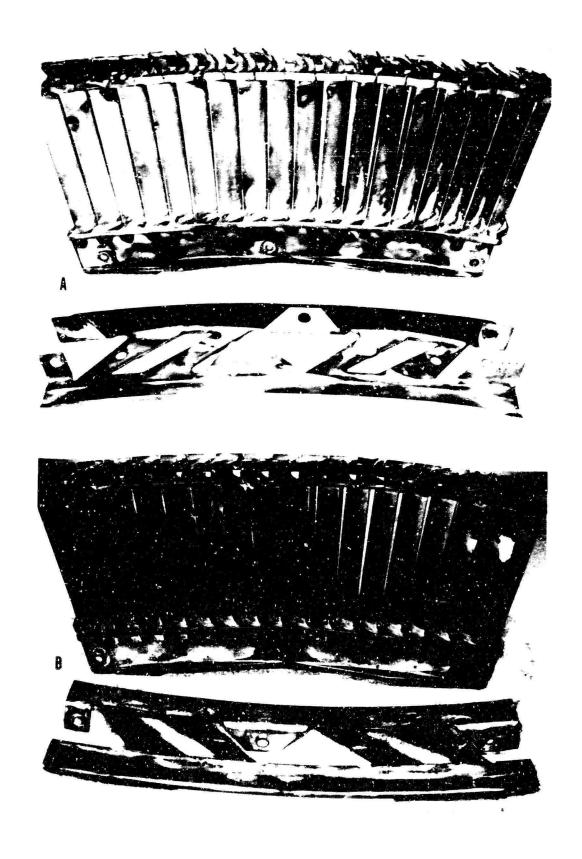
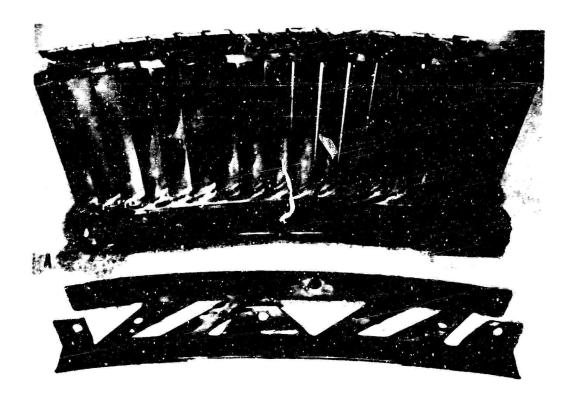


Figure II-41. Rotor

A. Carrier S/N 41 Forward Side (CASS 0757)

B. Carrier S/N 41 Aft Side (CASS 0742)

Note: Shroud Gouge, Airfoil Tears



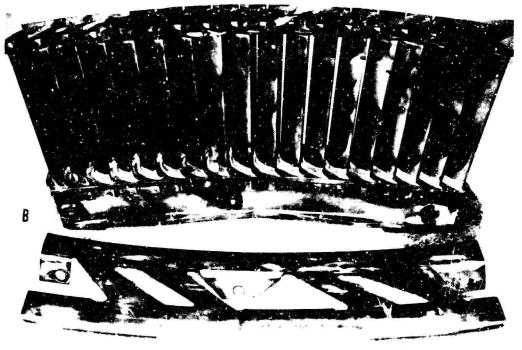


Figure II-45. Rotor

A. Carrier S/N 54 Forward Side (CASS 0751)
B. Carrier S/N 54 Aft Side (CASS 0752)

Note: Shroud and Airfoil Tears

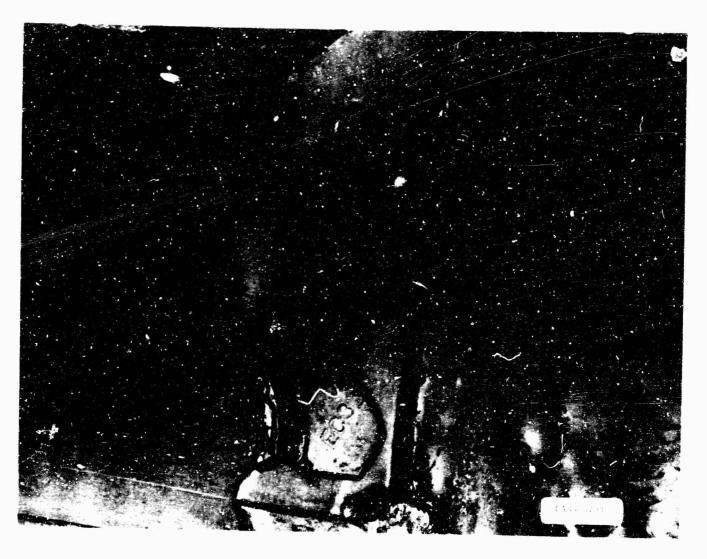


Figure II-46. Rear Frame
Missing Carrier Tab Lodged In
Insulation Blanket (CASS 0703)







Figure II-47. Rear Frame

- A. Cut In Insulation Blanket (CASS 0704)
- B. Cut In Insulation Blanket (CASS 0701)
- C. Puncture In Rear Frame Skin (Opposite B.) (CASS 0699)

Cause: Migration Of Carrier Tab

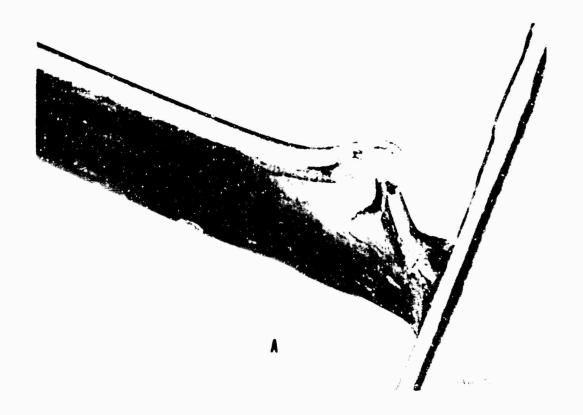




Figure II-48. Rear Frame

- A. Turbine Stator Tear (Carrier Tab) (CASS 0713)
- B. Fan Stator Cut (Washer, Not Part Of Test Vehicle, Found Lodged In Vane During FWT) (CASS 0711)

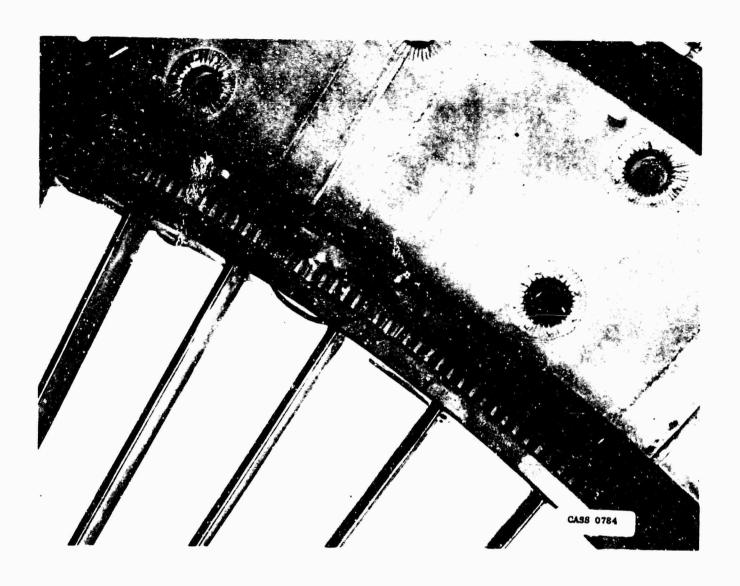


Figure II-49. Rear Frame
Bent Air Seal (From Migrating Carrier Tab) (CASS 0784)

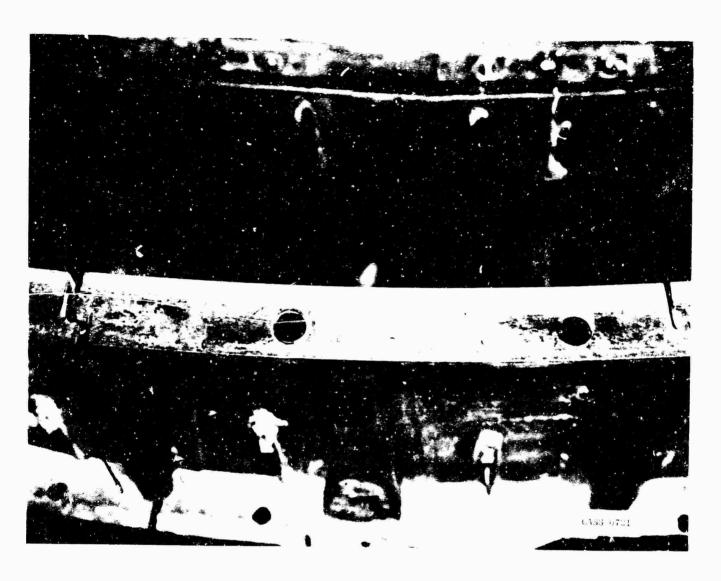
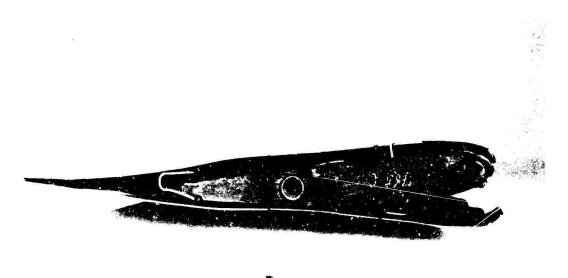


Figure II-50. Rear Frame
Thermal Break In Flange Relief Cut (CASS 0721)



C2111505

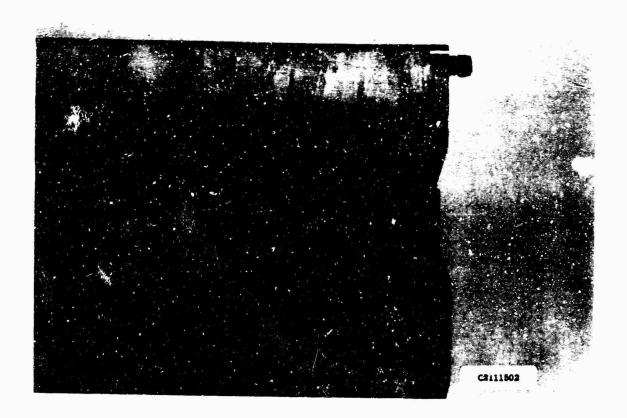


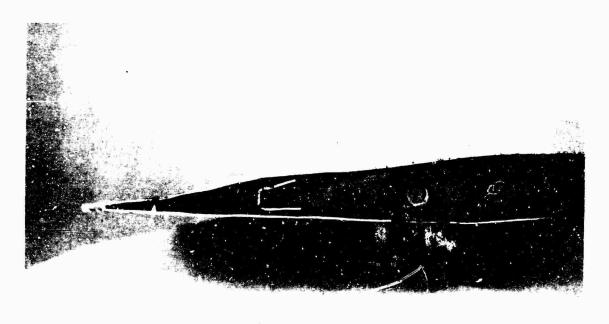
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Figure II-51. Exit Louvers

- A. Skin Failure At Load Bearing Eng Cap (C 2111505)
- B. End View Of A (C 2111504)





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Figure II-52. Exit Louvers

A. Failed Spot Welds (Undersize Nuggets (C 2111502)
B. End View Of A. (C2111501)

C2111501

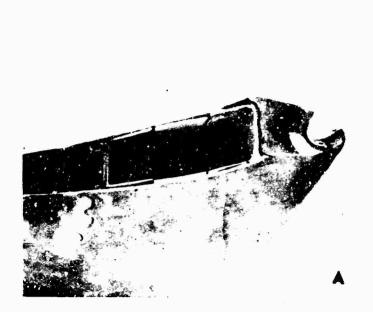






Figure II-53. Exit Louvers

- A. End Cap Weld Failure (Undercu Weld) #36
- B. End Cap Failure; Louver Dislouged During Operation (Failed Plug Welds) #23
- C. Thermal Buckling Adjacent To Turbine Stream #38 (3 Polaroids)

Note: Typical Failures



Figure II-54 Exit Louvers
Typical Spot Weld Failure and Resultant
End Cap Failure - #22



Figure II-55 Exit Louvers
Typical Spot Weld Repair

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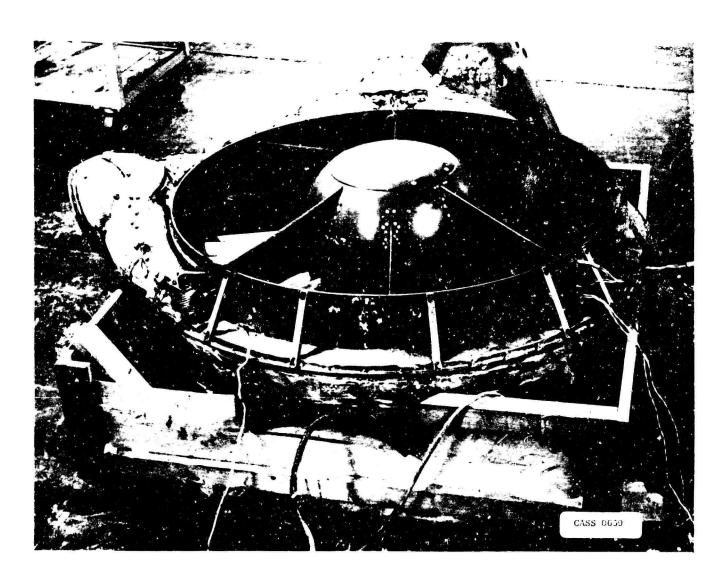


Figure II-56. Pitch Fan Assembly
Pitch Fan After Completing FWT

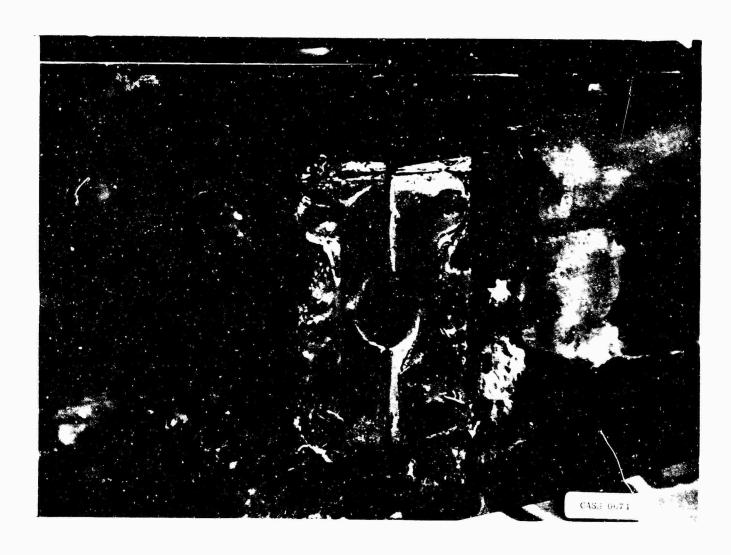


Figure II-57. Front Frame
Insulation Added To Frame Active Arc And Insulation
Pad Installed Between Frame And Scroll Center Mounts
To Correct Over-Temperature Condition (CASS 0674)
Note: Pad Installation Completed Without Fan
Disassembly

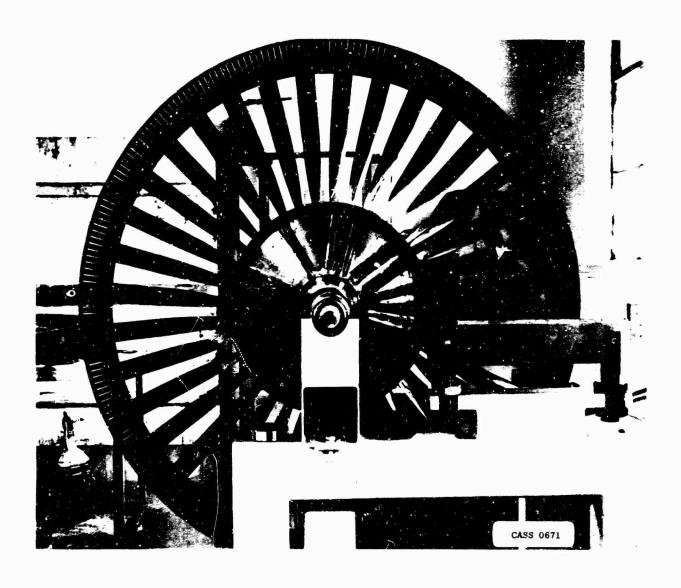


Figure II-58. Rotor
Rotor Sub-Assembly Prior To Disassembly After
Completing FWT



Figure II-59 Rotor
Roller Bearing Inner Race Burnished And
Discolored By Aft Grease Seal
Note: Light Groove In Beari: Path



Figure II-60. Rotor
Missing Carrier Anchor Nut (CASS 0683)
(Broken Loose At Disassembly)



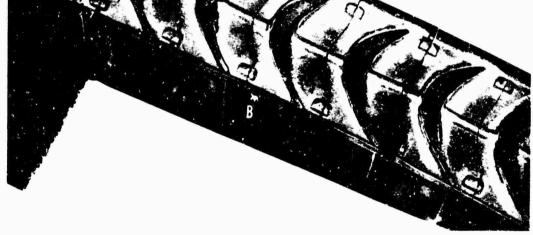


Figure II-61. Rotor
A & B. Missing Shroud Sections (CASS 0679, 0678)
Probable Cause: Braze Faults

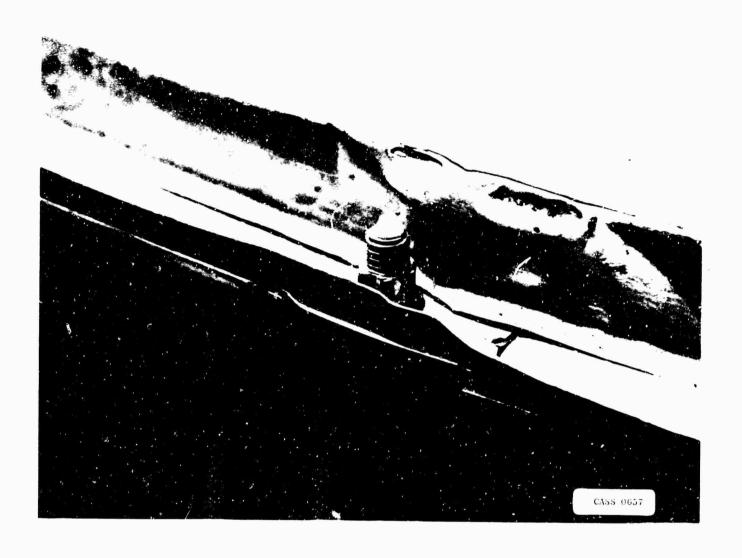
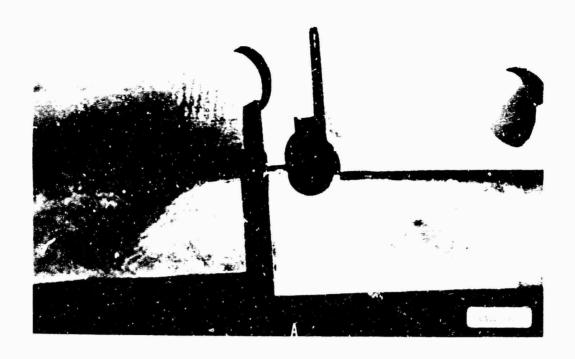


Figure II-62. Rear Frame
Insulation Cover Dents (CASS 0657)
Probable Cause: Damaged During Removal
From Test Facility



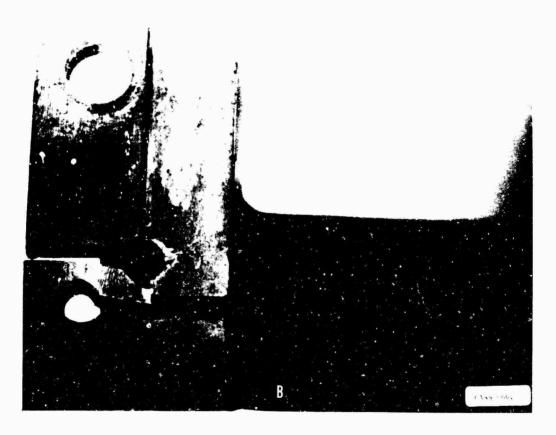


Figure II-63. Rear Frame
A & B. Thermal Distortions And Breaks In Flange
Relief Cut (Typical Of 5) (CASS 0660, 0661)



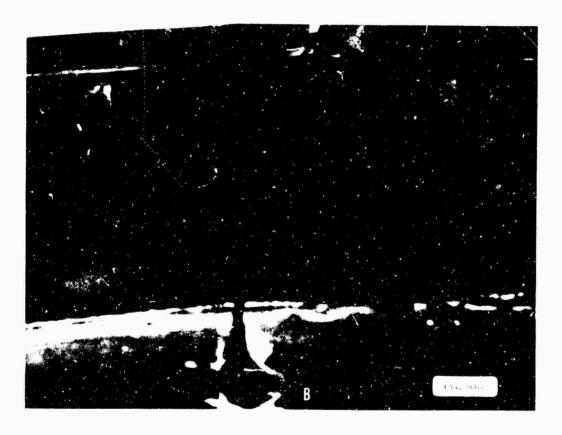


Figure II-64. Rear Frame
A & 3. Inner Skin Fatigue Failures At Expansion
Joint (CASS 0669, 0668)

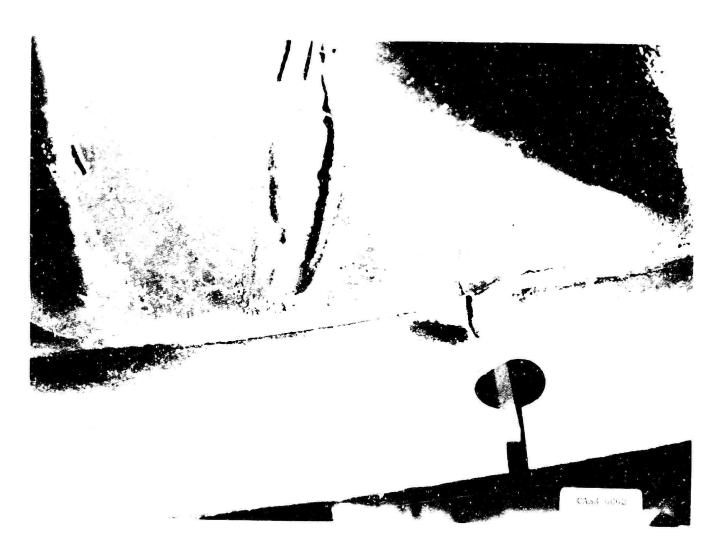


Figure II-65. Rear Frame
Thermal Cracks In Box Section Between Fan And Turbine
Stators (CASS 0662)
Note: Dent In Box Section Occurred In Fabrication



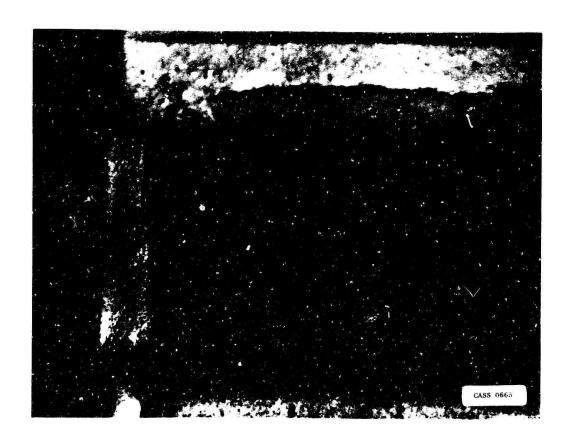


Figure II-66. Rear rrame

A & B. Crack In Braze Between Hat Section And Frame
Inner Wall (Inactive Arc) (CASS 0665)

INSPECTION CERTIFICATES
DIVERTER VALVE

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PURCHASE ORDER

	PART NUMBER 4012153-36071372Live
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PURCHASE ORDER

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NOTARY PUBLIC
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NOTARY PUBLIC
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LPP Fan Design Engineering

S/N 263 7/0# 2

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	INSPECTOR Lugar & Boun
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PART NUMBER 4012001-148
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Flight Research Program LPP Pan Design Engineering

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The General Electric Company, Cincinnati Aircraft Service Shop, hereby certifies that all parts indicated below have been fluorescent penetrant inspected by us and conform to all specifications indicated on the drawing, government specification, or as called for on purchase order.

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APPENDIX





Figure II-67. Pitch Fan Ducts

- A. Thermal Buckles (C 2110014)
- B. Buckling After Center Duct Guide Post Grew Out Of Guide Slot And Prevented Return To Nominal Position At Shutdown (C 2110105)

Note: Inspection After 29 Hours



Figure II-68. Crossover Duct
Internal View After 29 Hours Operation (C 2110108)



Figure II-69. Cross Duct
Bellows Distortion (After 29 Hours) (C 2110107)



Figure II-70. Cross Duct
Crack In Weld Affected Area (After 75 Hours)
Repaired In Place (C 2121033)



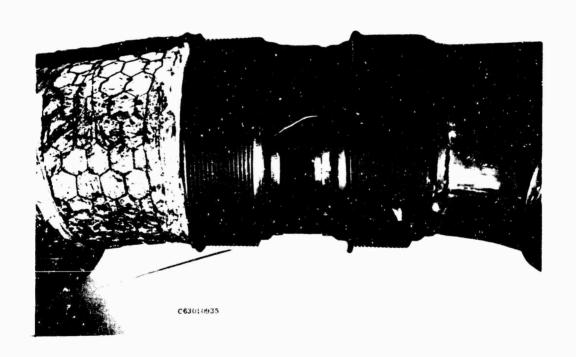


Figure II-71. Twist Buckle in .010" Pitch Fan Duct #2 At Pitch Fan.
Noted After 78 Hours Of Test Installation Prevented
Identifying Actual Time Of Incident





Figure II-72. Two Cracks In Crossduct #2 In Weld Affected Areas.
Noted After 78 Hours of Test. The Bottom Picture
Another View Of One Of The Cracks